

AMRL-TR-75-50 Volume 70



USAF BIOENVIRONMENTAL NOISE DATA

Volume 70 A-37B AIRCRAFT, NEAR AND FAR-FIELD NOISE

NOVEMBER 1975



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FOR THE COMMANDER

HENNINGE. VON GIERKE

Biodynamics and Bionics Division Aerospace Medical Research Laboratory

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data are reported for six locations in a wide variety of physical and psychoacoustic measures: overall and band sound pressure levels, C-weighted and A-weighted sound levels, preferred speech interference level, perceived noise

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PREFACE

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 723104, Measurement of Noise and Vibration Environments of Air Force Operations.

The author gratefully acknowledges Mr. John Cole for his assistance in preparing this report, Mr. Robert England for his assistance in acquiring the raw data, Mr. Henry Mohlman and Mr. David Eilerman of the University of Dayton for assistance in the mechanics of data processing and Mrs. Norma Peachey and Mr. Mike Patterson for assistance in typing and preparation of the graphics.



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INTRODUCTION

The USAF A-37B is a close air support and attack-type aircraft powered by two J85-GE-17A turbojet engines. The aircraft was manufactured by the Cessna Company and the engines by the General Electric Company.

This volume provides measured and extrapolated data defining bioacoustic environments produced by this aircraft during ground runup operations. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with ground runups of the A-37B aircraft.

This volume is one of a series published by the Aerospace Medical Research Laboratory (AMRL) under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and aerospace ground equipment. The far-field, community-type noise data in the handbook describe the noise produced during *ground operations* of aircraft, aerospace ground equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. Volume 2 provides a method and data for adjusting the handbook's far-field noise data, which are for standard meteorological conditions (15 C temperature, 70% rel humidity, 0.760 meters Hg barometric pressure), to derive comparable data for other meteorological conditions. Refer to Volumes 1 and 2 (references 1 and 2) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., inflightg-flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published and is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of each updated index.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; AUTOVON 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application, AMRL-TR-75-50 (1), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975

^{2.} Cole, John N., USAF Bioenvironmental Noise Data Handbook, Volume 2: Procedure to Evaluate Effects of Non-standard Meteorological Conditions on Far-Field Noise, AMRL-TR-75-50 (2), AMRL, WPAFB, OH, 1975

NEAR-FIELD NOISE

MEASUREMENTS

AMRL acquired near-field noise data on the A-37B aircraft during ground runup operations of its turbojet engines. For these tests the aircraft was located on a concrete runup pad at Edwards AFB, CA, with no significant reflecting surfaces in the vicinity except the ground plane. Table 1 gives the surface meteorological conditions and the two engine/power conditions. The groundcrew chief selected power conditions and near-field locations generally used during routine maintenance or engine runup for preflight checks.

At each near-field location a test engineer randomly moved a hand held microphone in and around each location, probing all areas where a crew member's head would normally be located. He recorded all of the noise samples on magnetic tape. During analysis of each sample, he determined the root-mean-square sound pressure using a 4- or 8-second integration time to derive a power-averaged level for each location. Figure 1 shows the six near-field locations where ground crew are usually located for maintenance and/or pre-flight checkout operations. Estimates of noise levels at other locations in the near-field are difficult since the noise source is spatially distributed, i.e., not a point source. The noise levels at near-field locations can vary widely depending upon relative distances from each noise source (intake noise, exhaust noise, panel resonances, internal engine noise through the engine wall, etc.).

Table 1 lists the numeric/alphabetic designators used on the data pages in this report to identify the measurement locations and test conditions. For example, the designator 1/A means ground crew location 1 and test condition A.

RESULTS

The measured data presented in Table 2 define the sound pressure levels (SPL) produced by the A-37B aircraft at the six ground crew locations. This table includes the overall, 1/3 octave band, and octave band levels. From these data one can calculate the variety of measures given in Table 3, which are widely used to assess the effects of noise on personnel and their performance.

All near-field data are for the meteorological conditions at the time of test but are valid for all typical airbase meteorology because of the short sound propagation distance involved.

TABLE 1

MEASUREMENT LOCATIONS AND TEST CONDITIONS FOR NEAR-FIELD NOISE MEASUREMENTS

A-37B Aircraft, Ground Runup, Edwards AFB, CA 25 September 1972

Ground Crew Location 1 Engine #1 Start Engine #2 Start 2 Marshall 3 Wheel Chock Pull Flap Check 5 Hydraulic Leak Check 6 Aircraft Engine Operation Engine #1 Idle Power A Both Engines Idle Power B MeteorologyTemperature Bar Pressure 10.0 C 0.704 M Hg 87 % Rel Humidity Wind — Speed — Direction 3.1 M/Sec (6 Kt) 220 Deg.

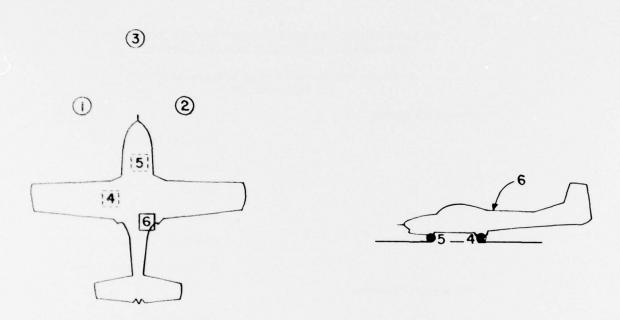


Figure 1. Near-Field Measurement Locations at Pad 17, Edwards AFB, CA

FAR-FIELD NOISE

MEASUREMENTS

AMRL acquired both near and far-field data during a 1-2-hour test period, thus keeping similar meteorological conditions. Figure 2 shows the ground runup pad, ground cover, aircraft orientation and the 19 microphone measurement sites on a semicircle. The center of the 75 meter radius semicircle used in surveying the J85-GE-17A engines was on the ground directly below the intersection of the aircraft's centerline and the plane passing through both engines exhaust-nozzle exits.

Table 4 provides cockpit readouts of engine characteristics (% RPM, fuel flow, etc.) for each power setting used in the far-field tests. Also listed in this table are the surface meteorological conditions during data acquisition.

All microphone measurement sites are in the acoustic far-field of the source where the sound wave-fronts spherically diverge and the noise source may be regarded as a point source.

A portable microphone/tape-recorder system was used to sequentially record the noise at each far-field location. The microphone was attached to a hand held pole, pointed at the source (0° angle of incidence) and vertically scanned from 0.5 to 3 meters for a period of 5-10 seconds during data acquisition at each microphone location. These samples were then time-integrated to derive a root-mean-square sound pressure level. Vertical scanning and time-integrating together reduce anomalies frequently present in data acquired by a fixed height microphone.

RESULTS

Table 5 lists the overall and 1/3 octave band SPL measured at the far-field locations under meteorological conditions at the time of the test. Data in all other figures and tables are based on these levels. These data were normalized to 100 meters distance and standard meteorological conditions (15 C temperature, 70% relative humidity, 0.760 meter Hg barometric pressure) and used to derive the graphic data in Figure 3 which provides a compact summary of the far-field noise characteristics of the A-37B aircraft in a standard format.

Figure 4 and Table 6 present two basic acoustic measures, the acoustic power level and the directivity index, respectively. The acoustic power level describes the power radiated by the source as a function of frequency. The directivity index is a standard acoustical engineering measure that describes the geometric way in which the source radiates this power as a function of both frequency and angle from source. These basic source measures are primarily of interest for acoustical engineers and noise generation/control specialists.

Figures 5 through 11 are sets of equal noise contours describing seven different measures of noise as a function of angle and distance from the source for standard day meteorology. They are respectively, overall sound pressure level, C-weighted sound level, A-weighted sound level, perceived noise level, speech interference level, permissible exposure times for personnel and octave band sound pressure levels.

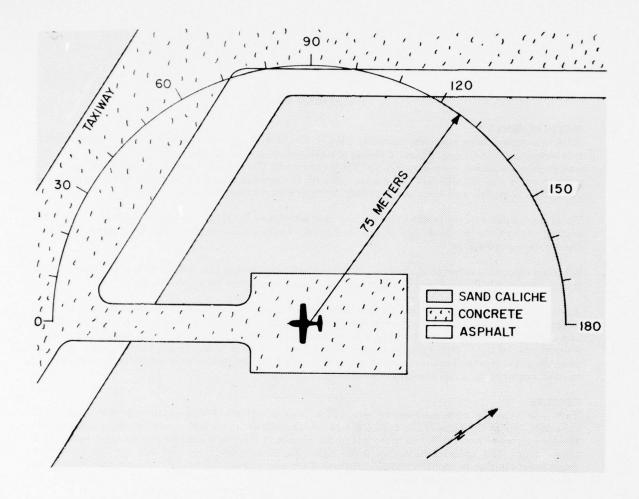


Figure 2. Far-Field Measurement Locations at Pad 17, Edwards AFB, CA

Data excessively influenced by spurious background/electronic noise were eliminated from all figures and tables. No data are presented at the 170 and 180 degree locations for the higher power settings because of turbulent air flow behind the aircraft.

Test personnel performed noise surveys during quiet periods when the background noise was minimal, e.g., early in the morning when no other aircraft or engine test stands were operating. Data eliminated because they were near the background/electronic noise were generally not significant because the levels were so low (e.g., Table 5 and Figure 11 at idle power).

Volume 2 of the handbook describes the influence of meteorology on far-field noise environments, and provides, if required, the factors necessary to adjust the handbook's standard meteorological day data.

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60	96	96	46	87	104	104	102	
125	9.6	100	101	95	103	108	109	
250	96	100	66	91	105	106	107	
500	37	9.8	26	96	104	105	108	
1000	101	101	103	91	105	106	106	
2000	107	104	103	66	108	111	106	
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BASED ON CALCULATED SPL SPECTRUM UNDER PROTECTIVE DEVICE. ADDITIONAL EAR PROTECTION REQUIRED.

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TABLE 4

TEST CONDITIONS FOR FAR-FIELD NOISE MEASUREMENTS

A-37B Aircraft, Ground Runups, Edwards AFB, CA 20~&~25 September 1972

Aircraft Engine Operation

Idle	Both	Engines
	46%	RPM NC (Core Speed)
	355	C EGT (Exhaust Gas Temperature)
		LBS/HR FF (Fuel Flow)

85% Runup	Both Engines
	85 % RPM NC
	490 C EGT
	1250 LBS/HR FF

Military	#2 (Right) Engines
	100 % RPM NC
	574 C EGT
	2250 LBS/HR FF

Meteorology (85% & Mil)

Temperature	10.0 C
Bar Pressure	0.704 M Hg
Rel Humidity	87 %
Wind — Speed	3.1 M/Sec (6 Kt)
- Direction	220 Deg

(Idle)

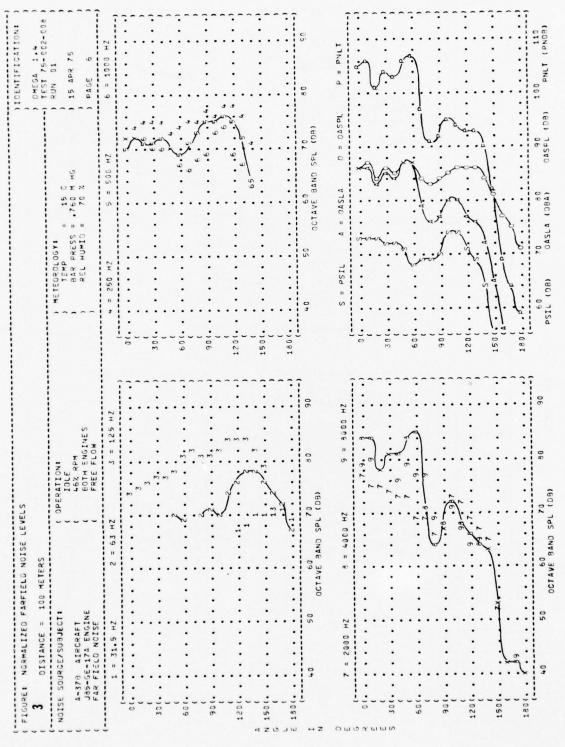
Геmperature	15.0 C
Bar Pressure	0.705 M Hg
Rel Humidity	59 %
Wind — Speed	Calm

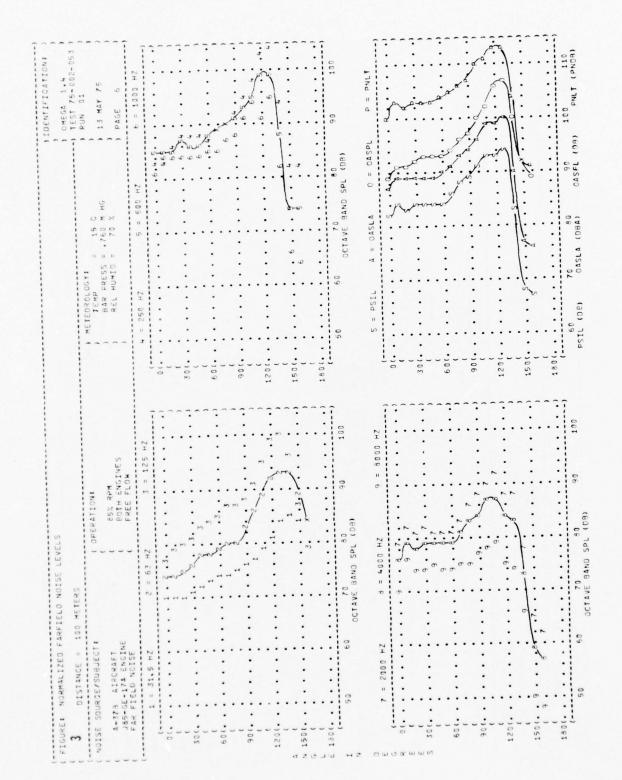
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~	99					99	90	63	9	68	69	69	20	68		9	•				
8.0	9					92	99	69	19	69	99	70	72	19		n	~				
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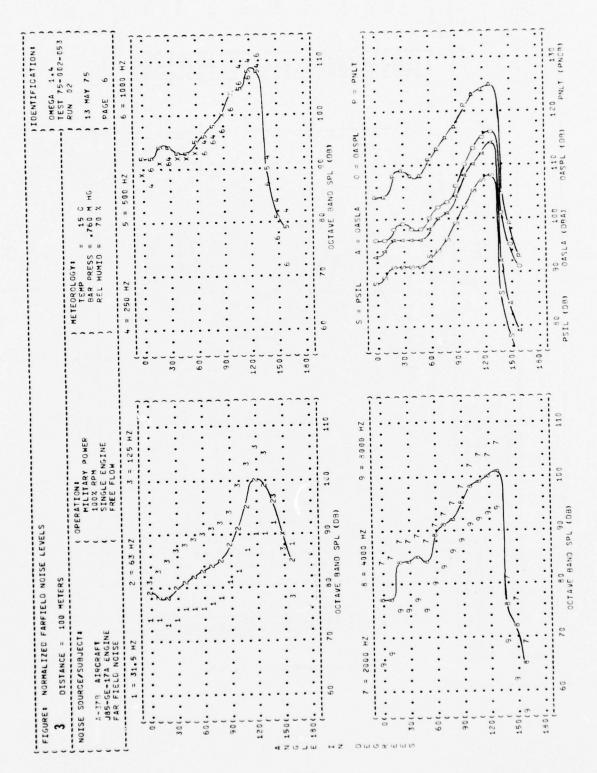
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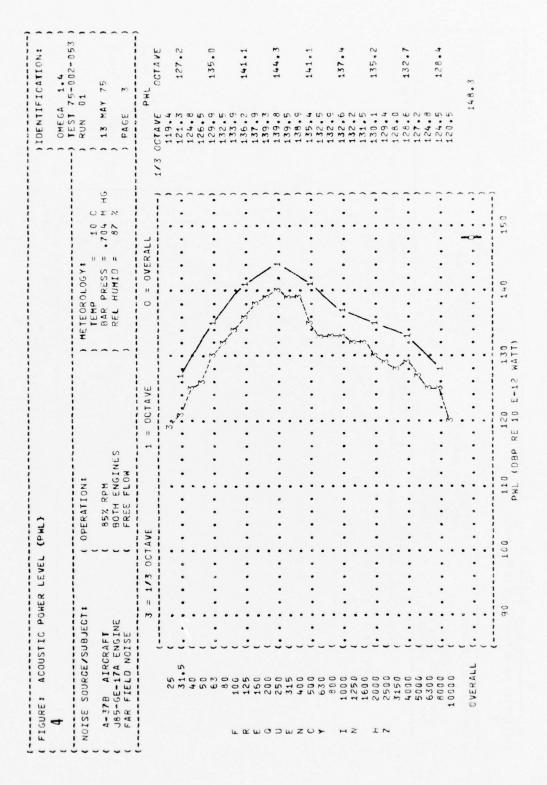
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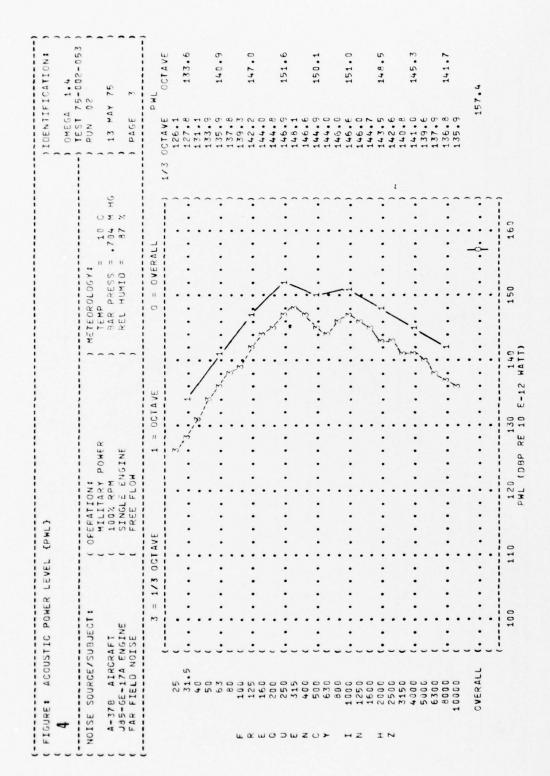






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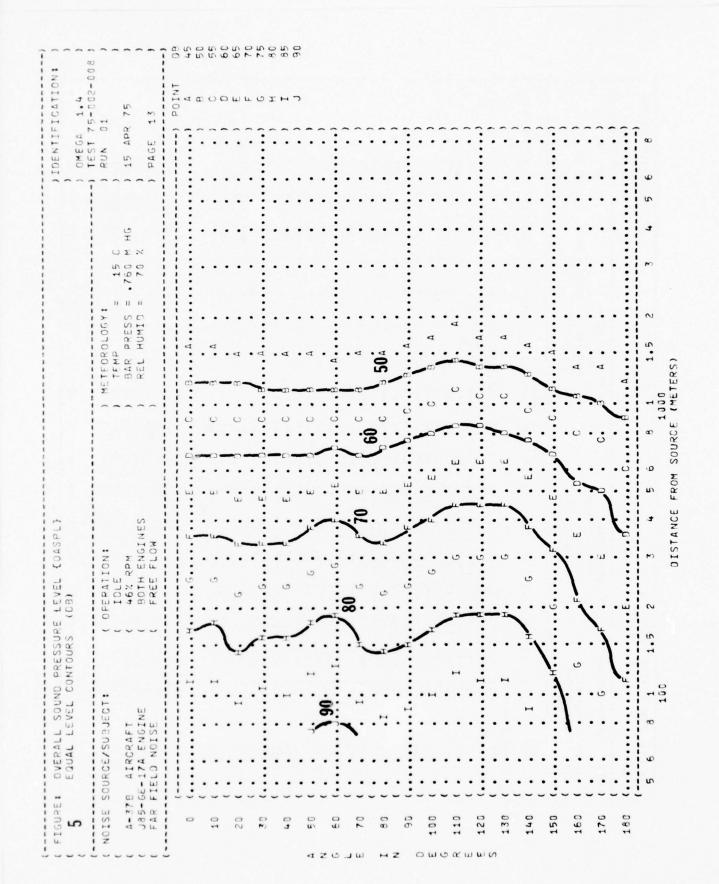


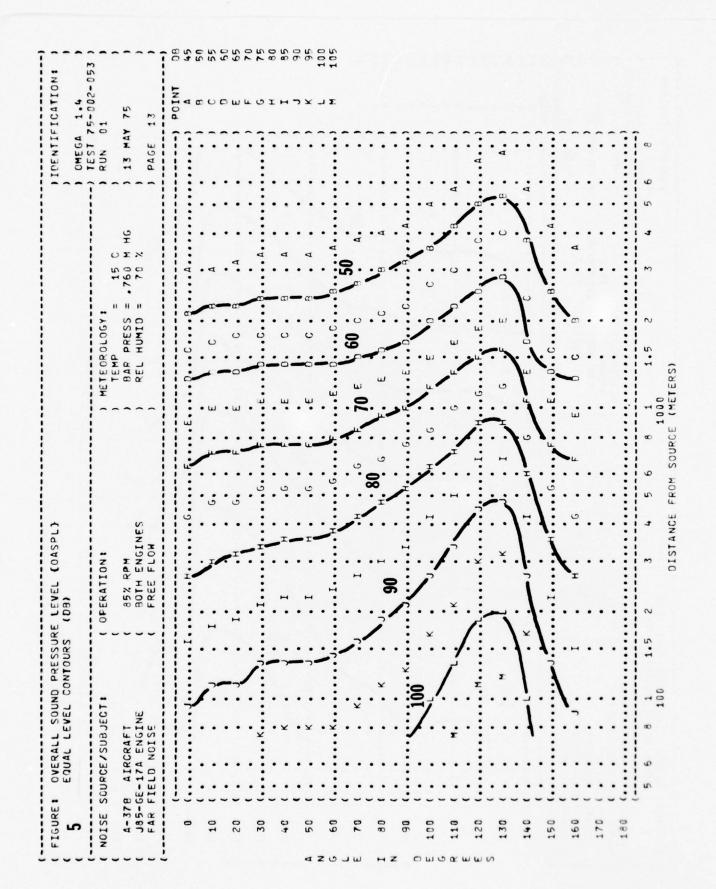


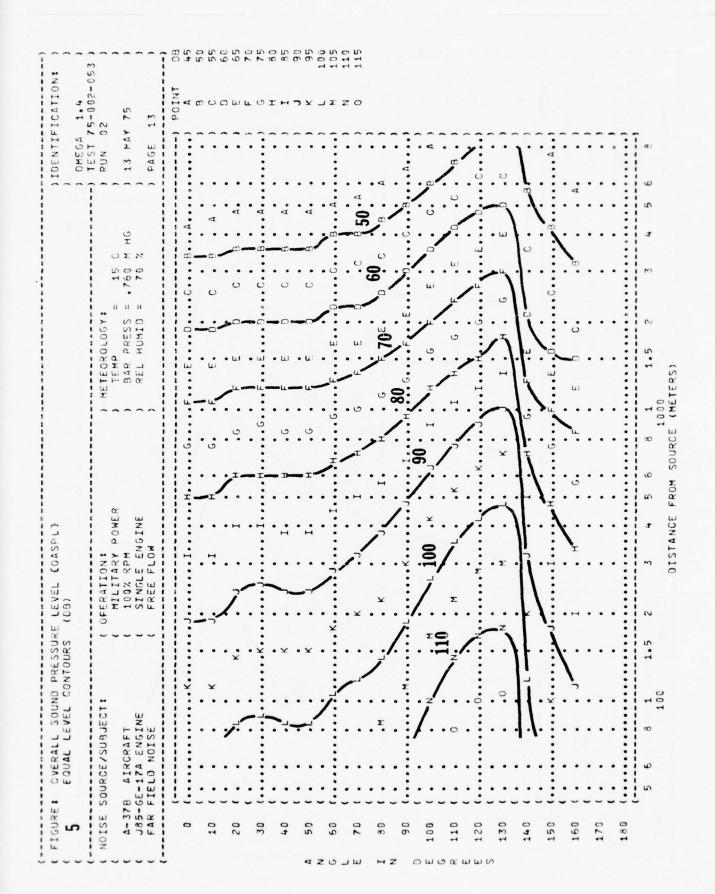
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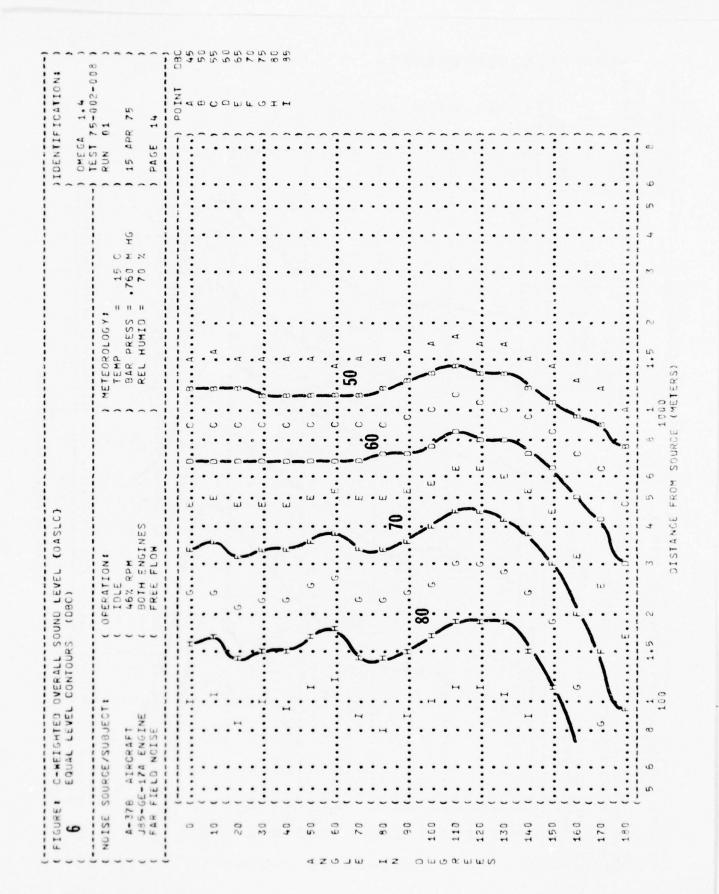
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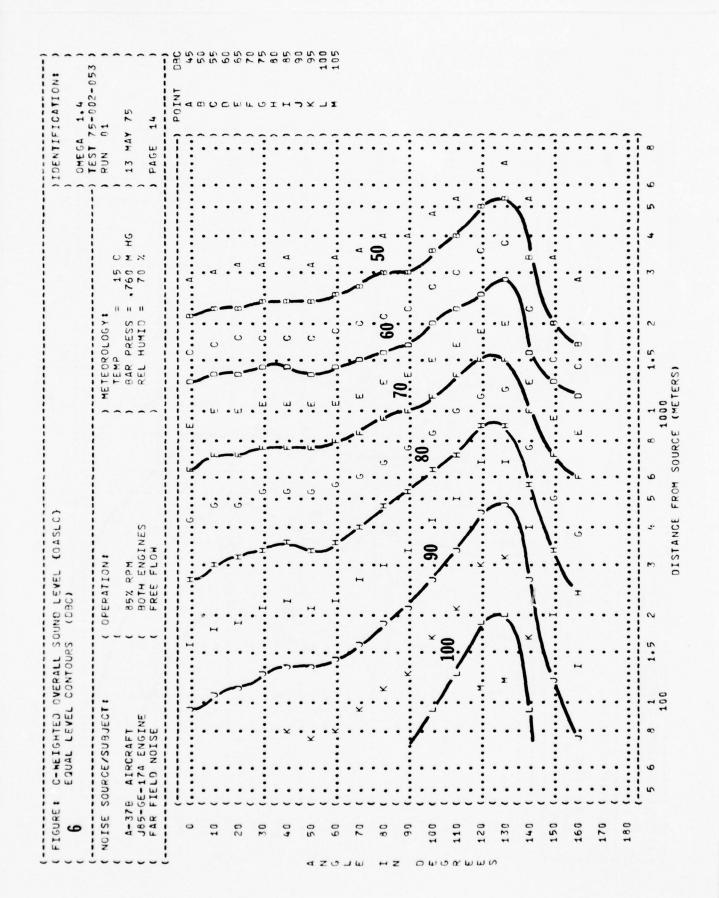
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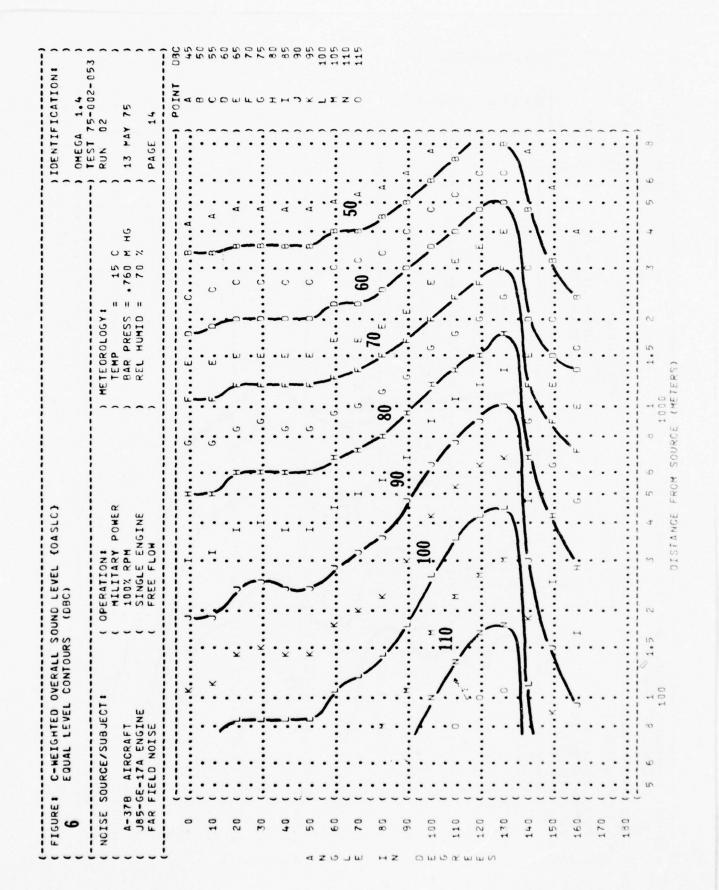


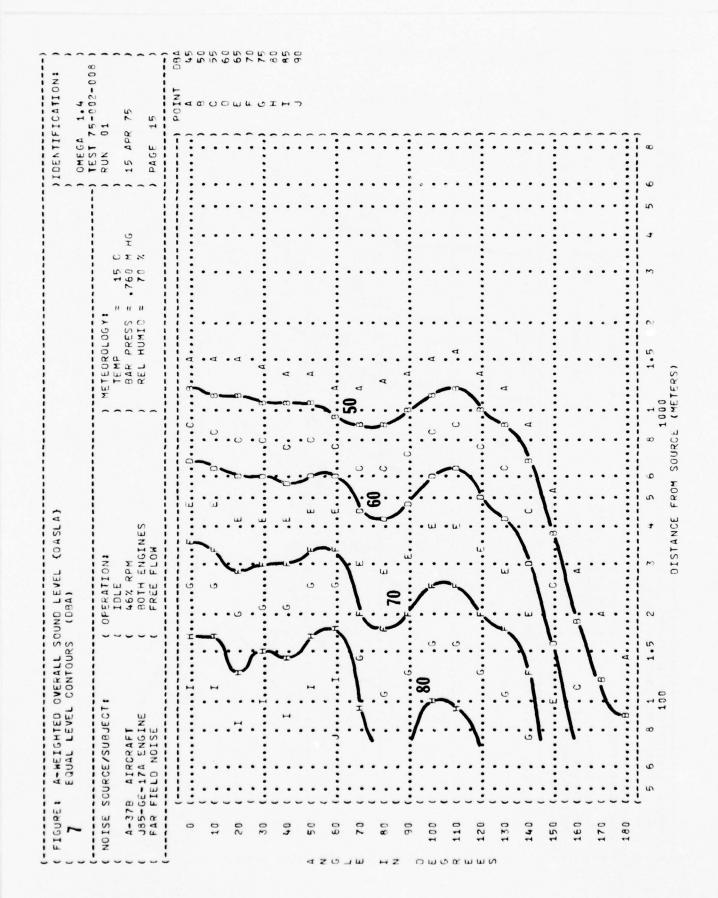


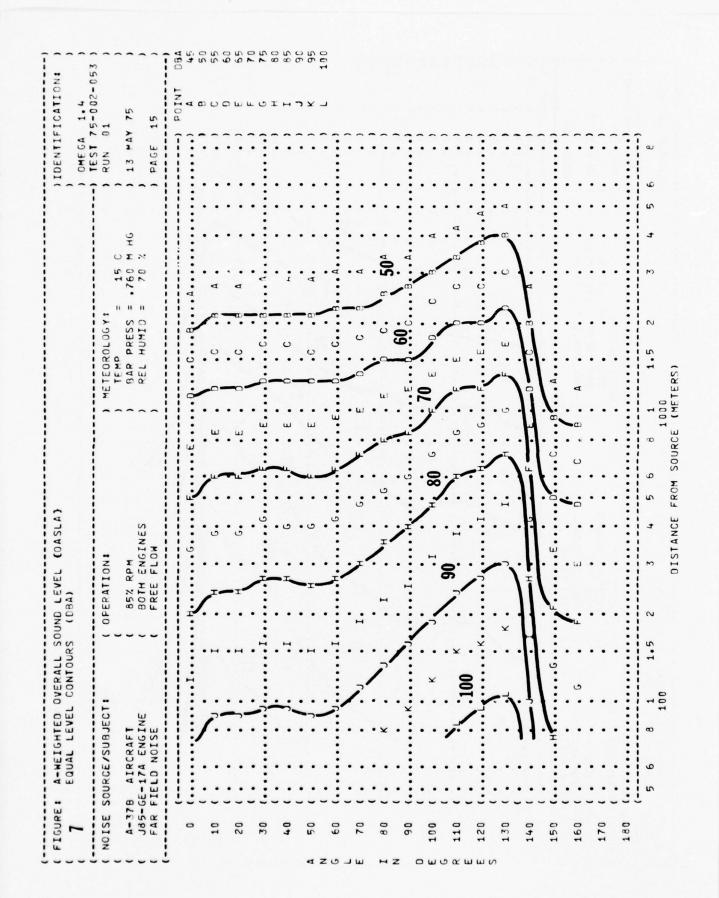


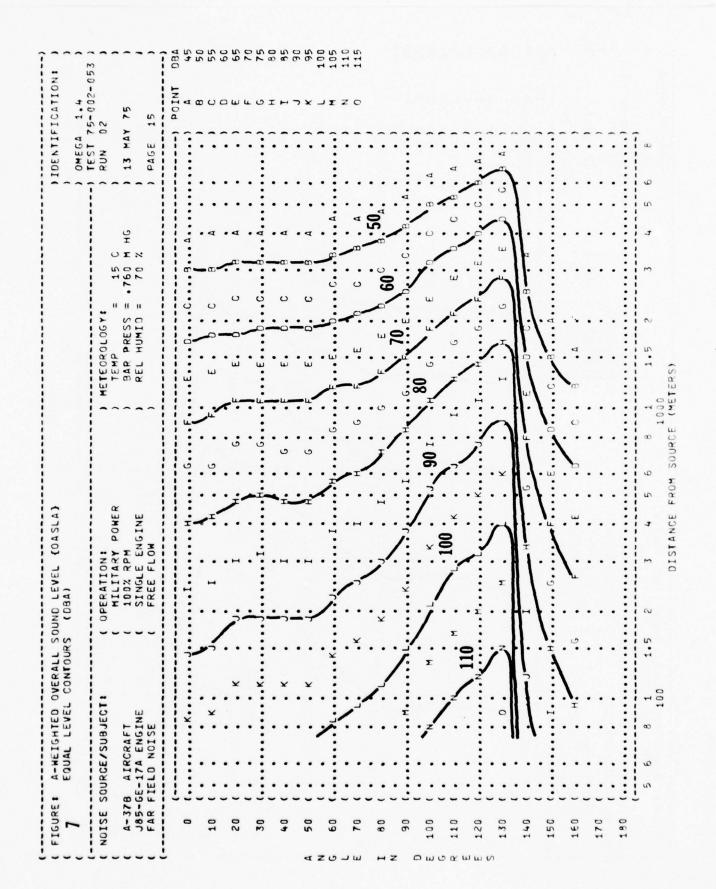


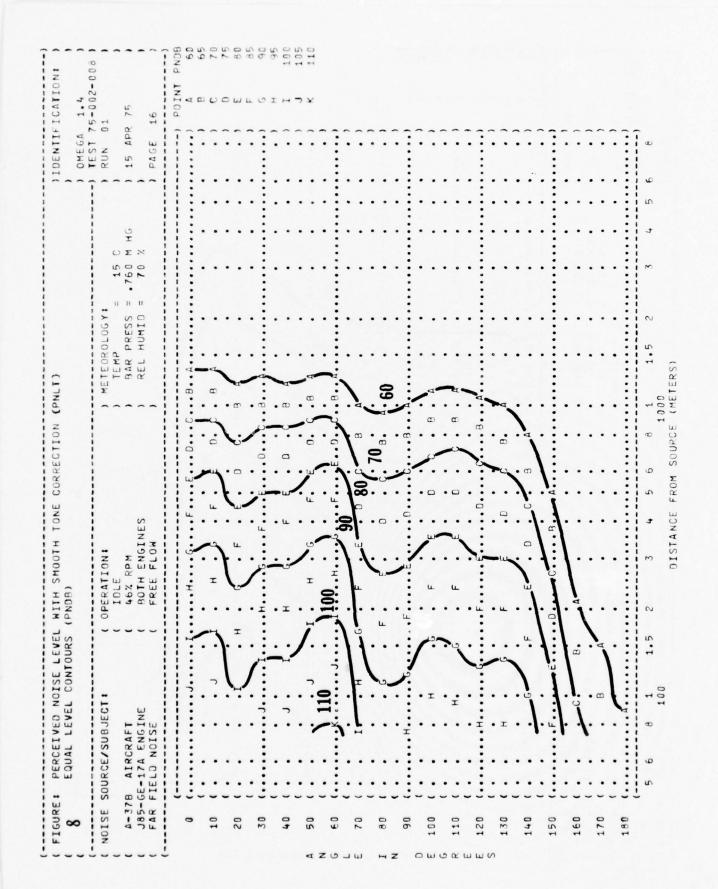


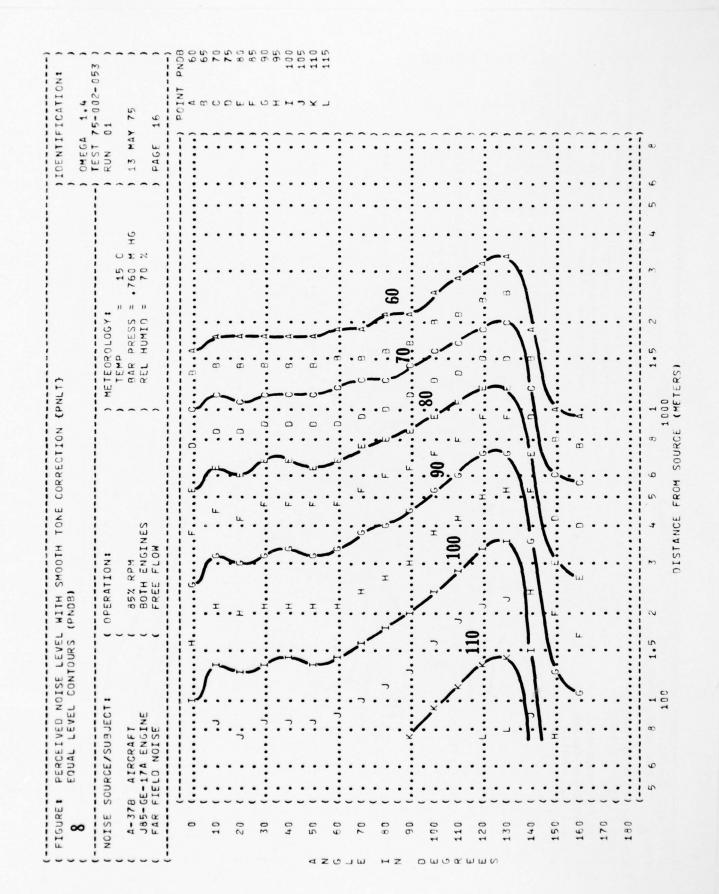


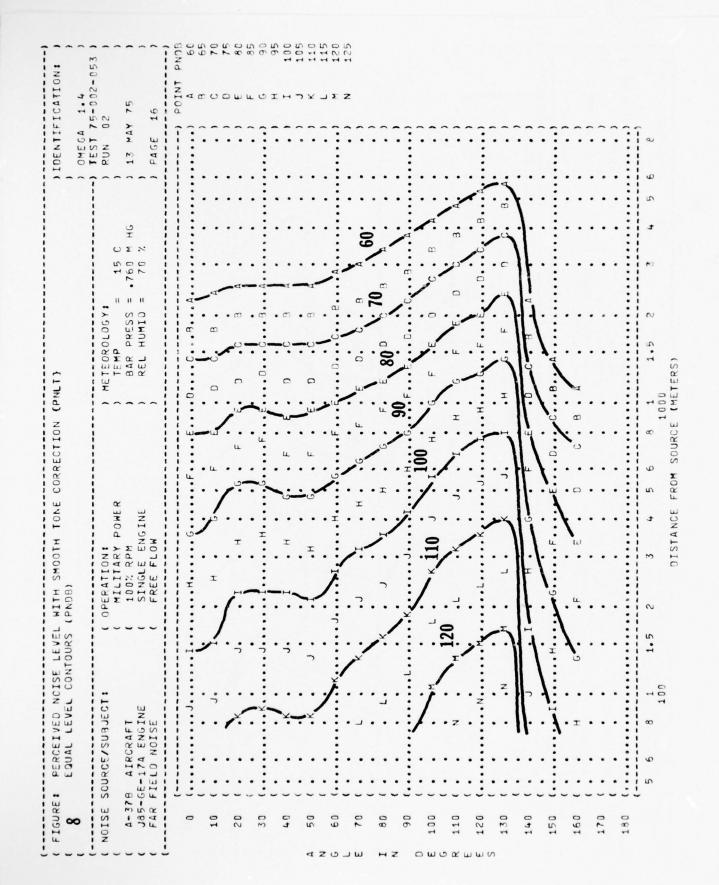


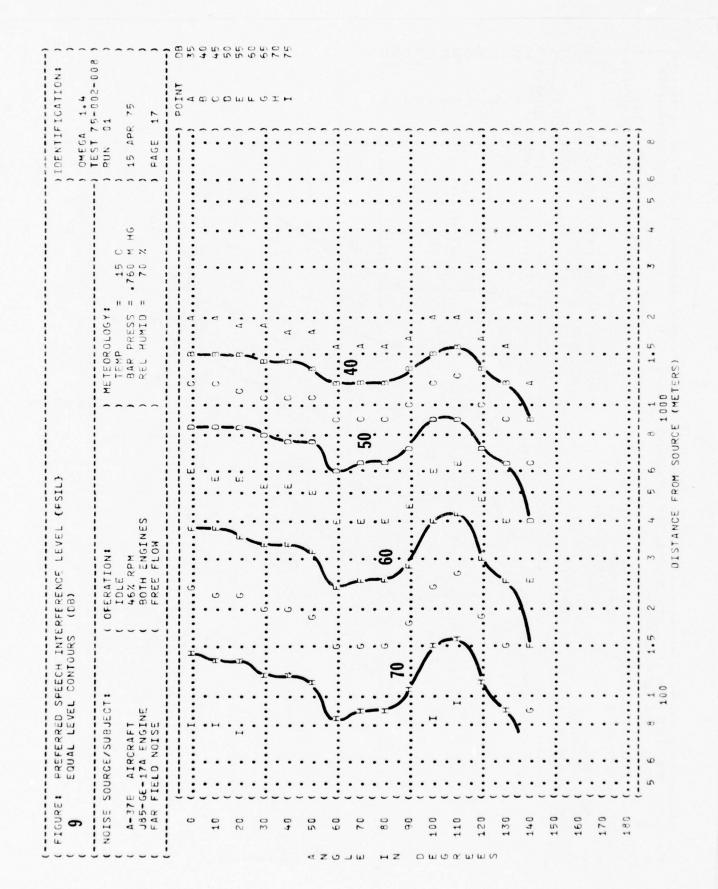


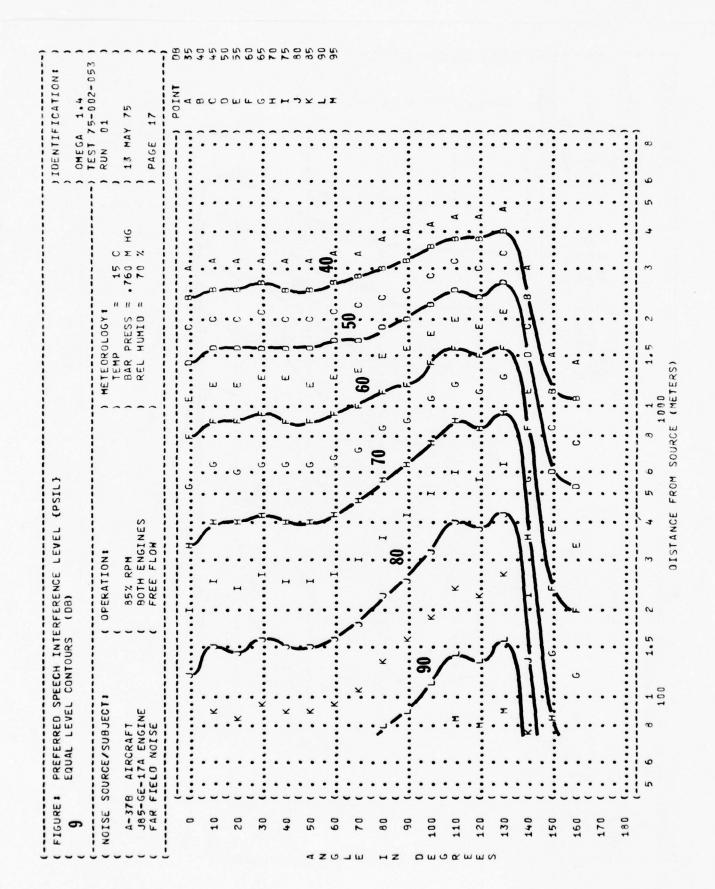


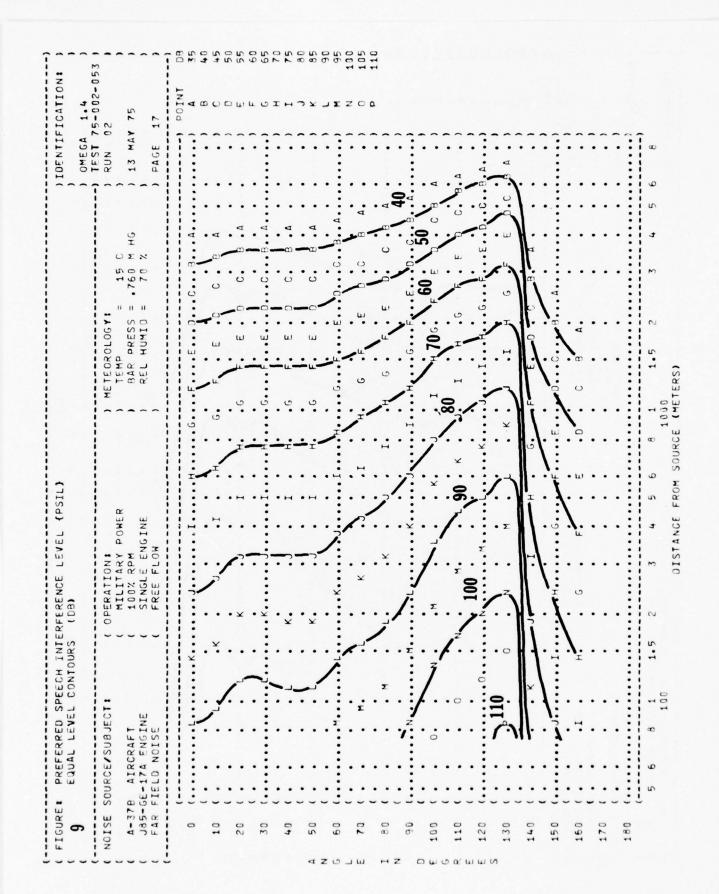












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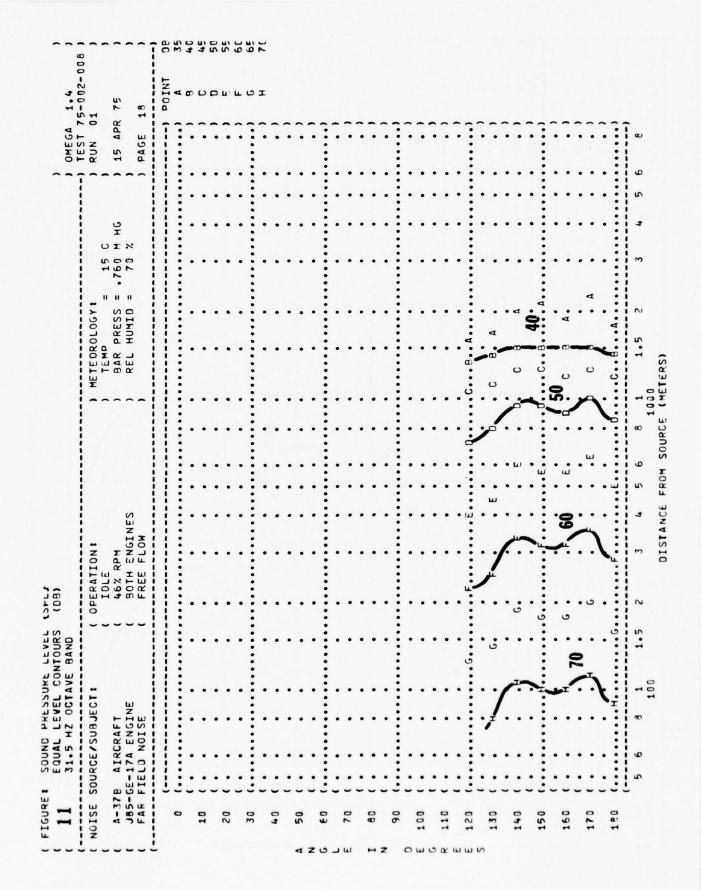
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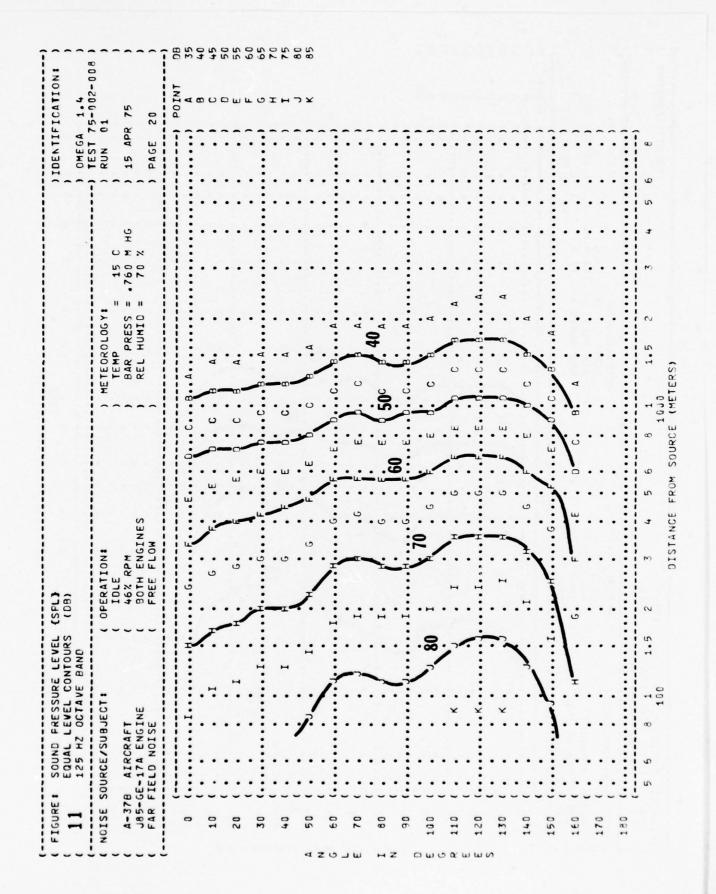
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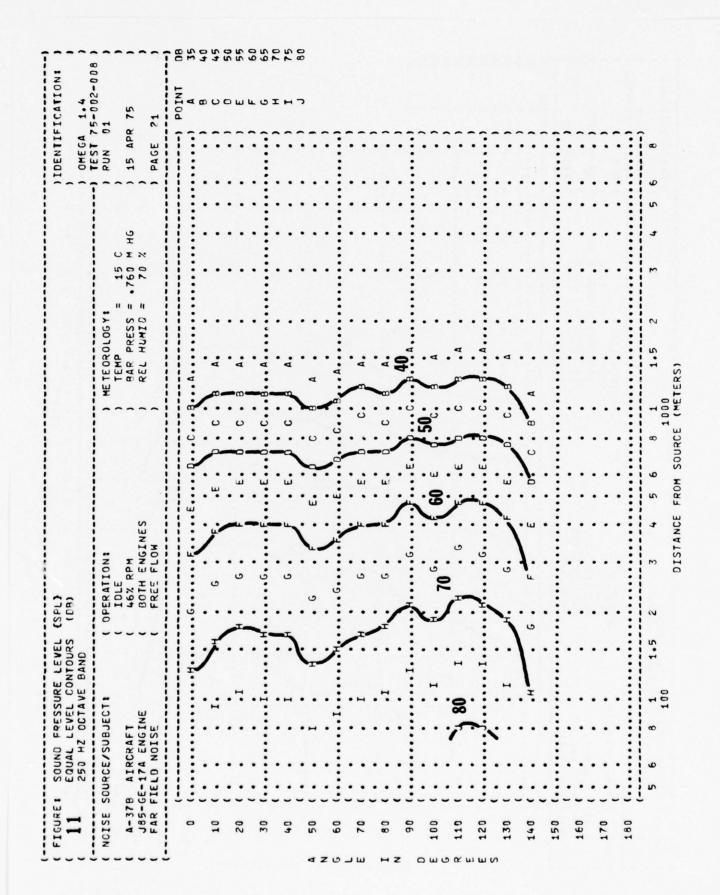
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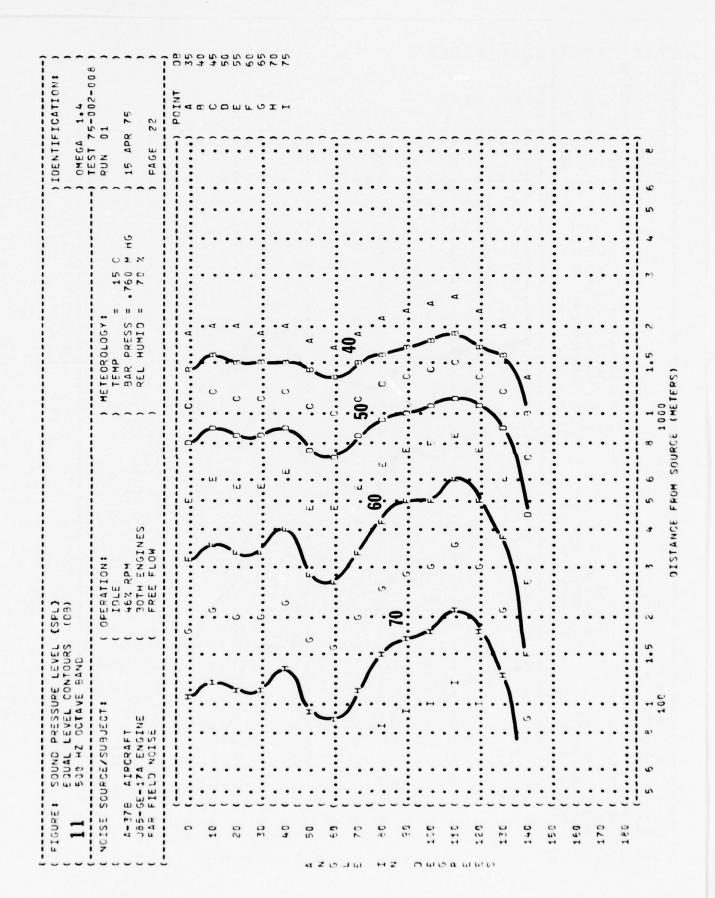
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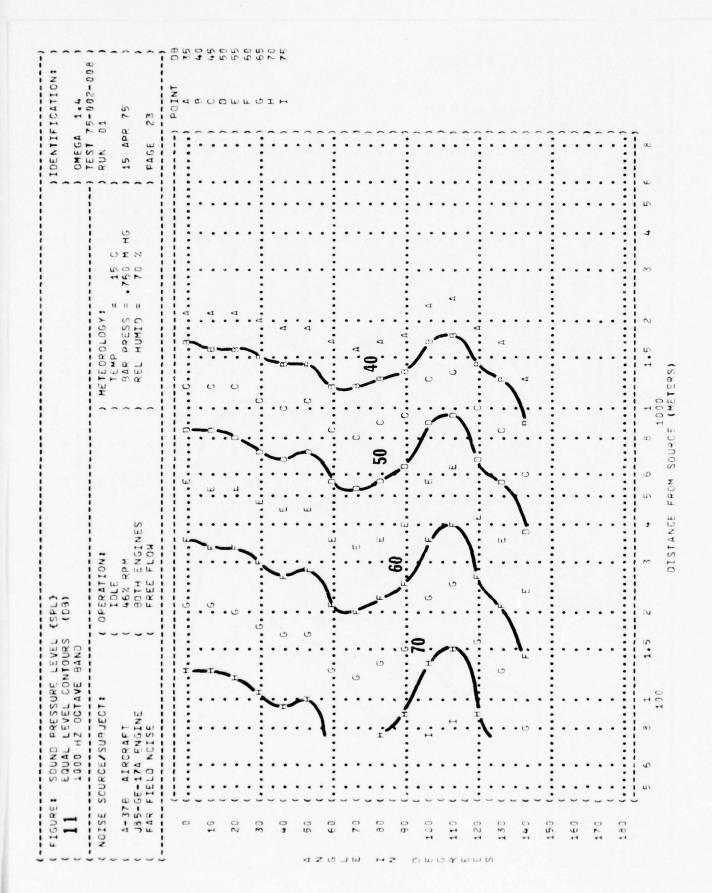


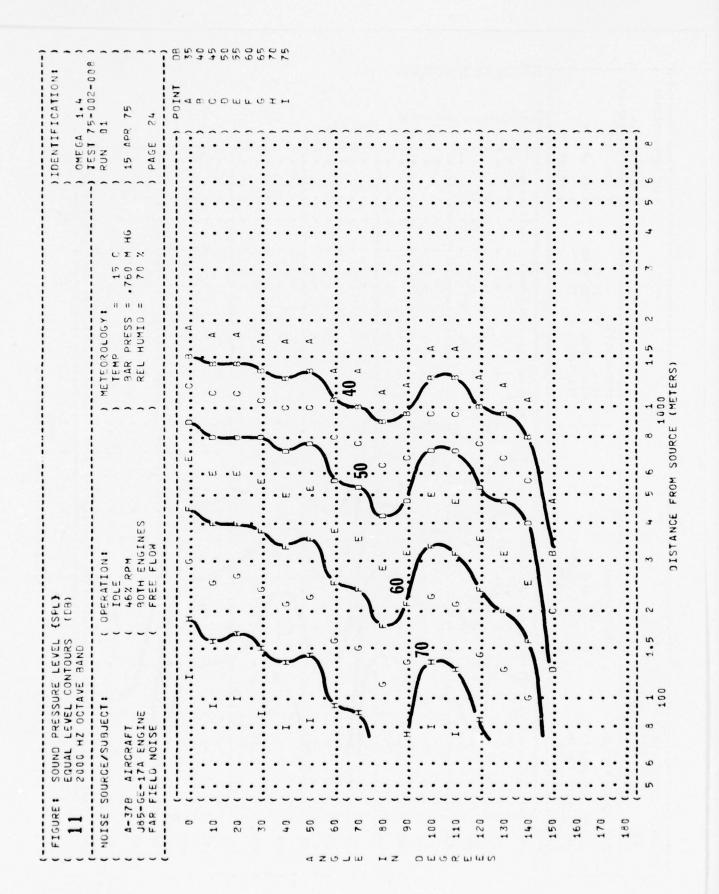
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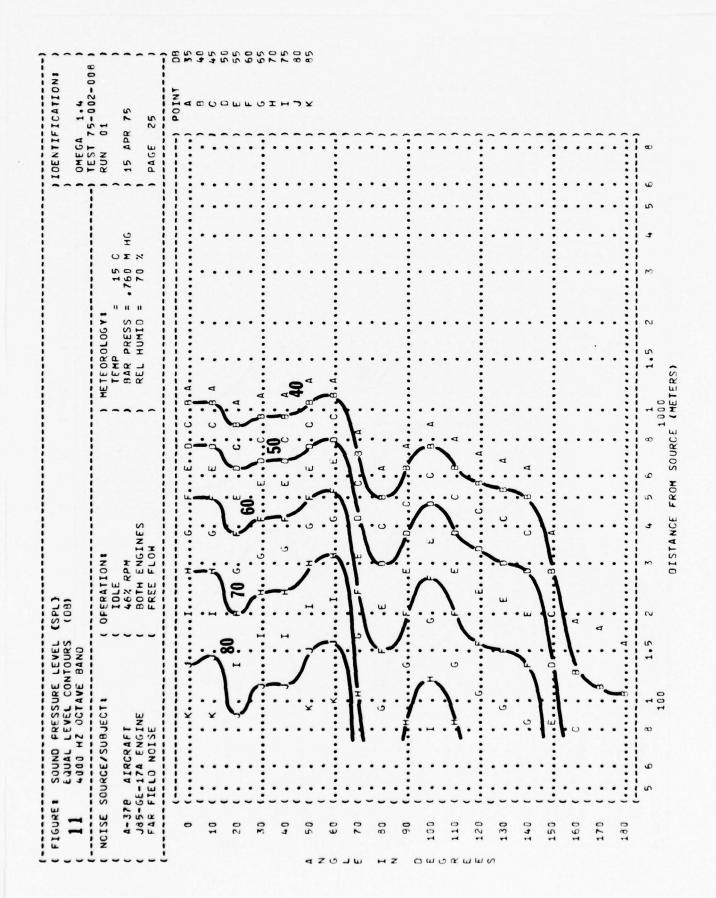






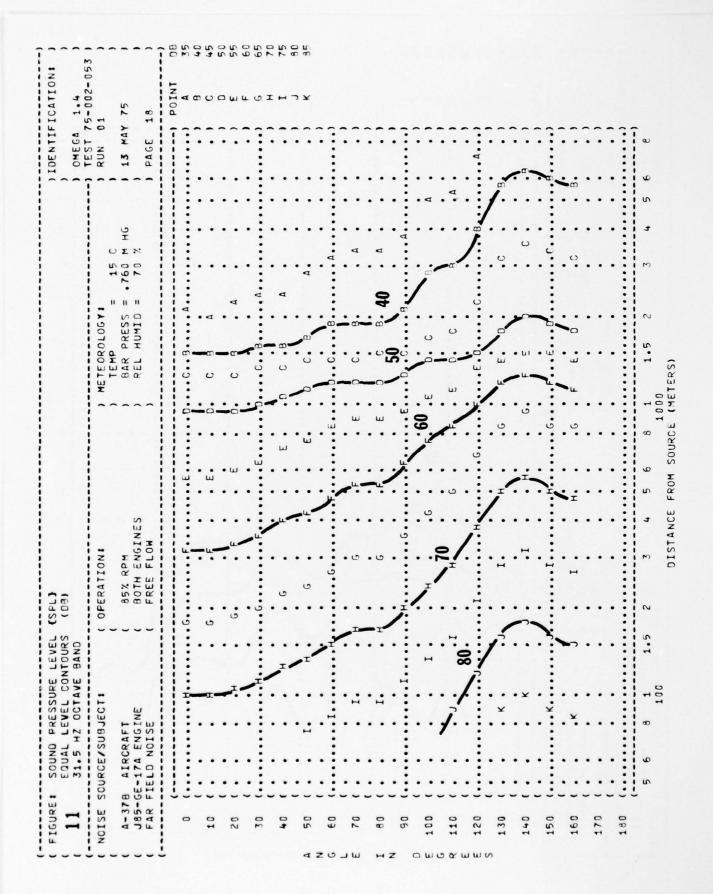


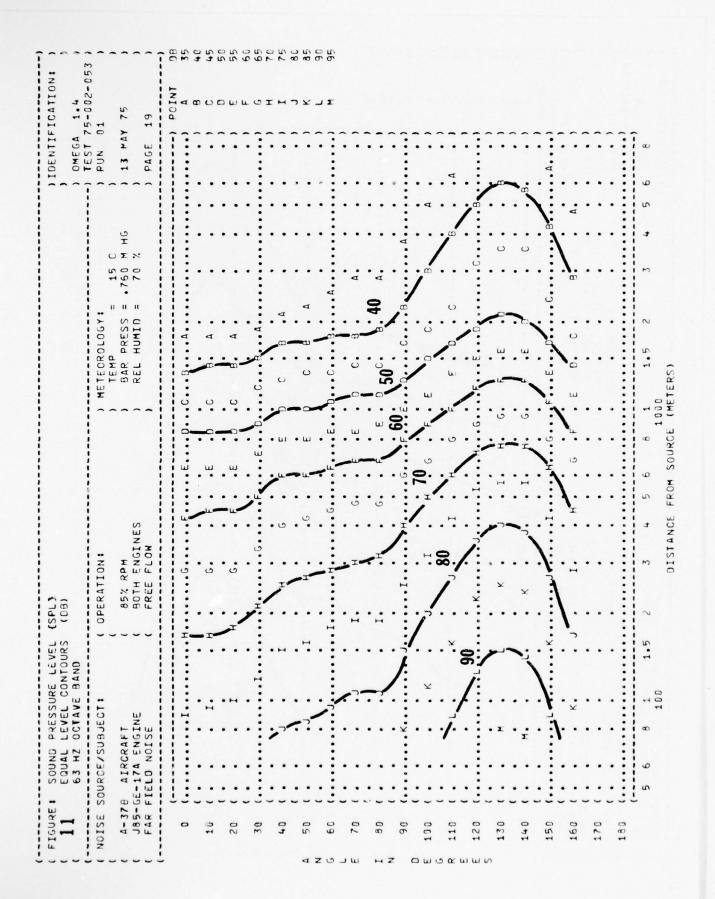


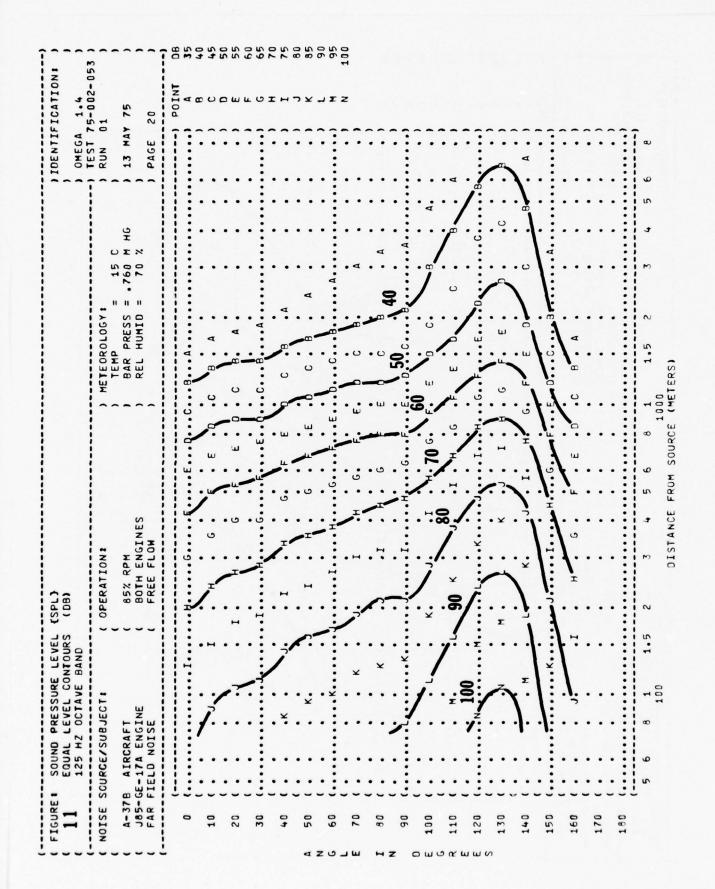


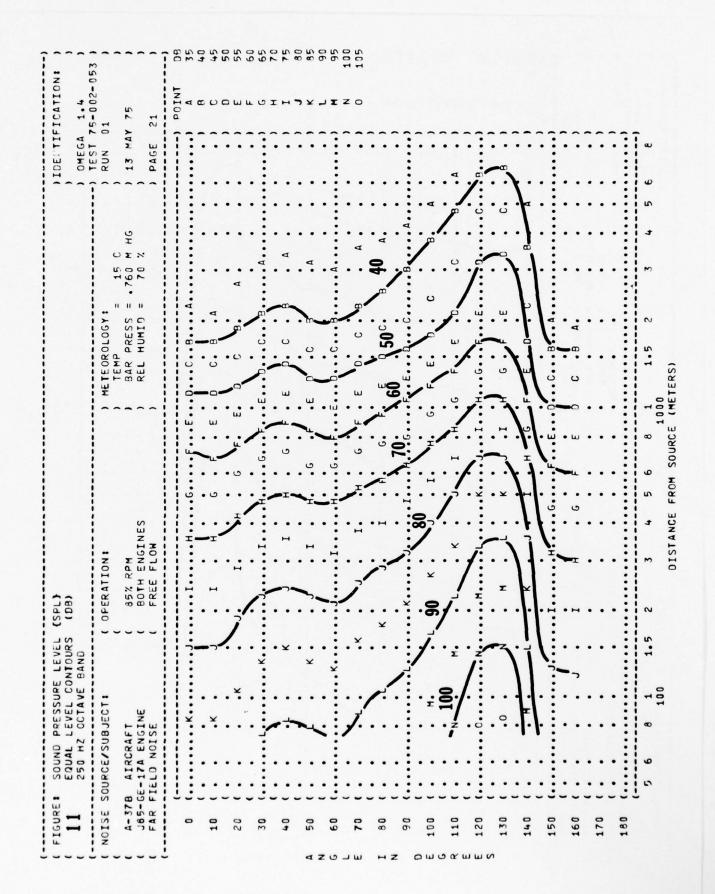
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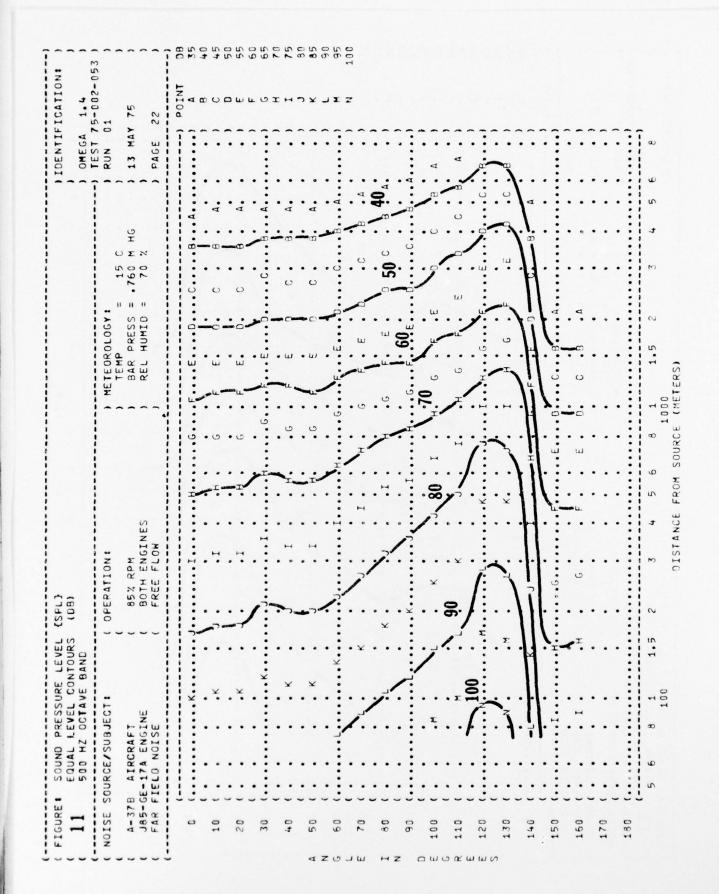
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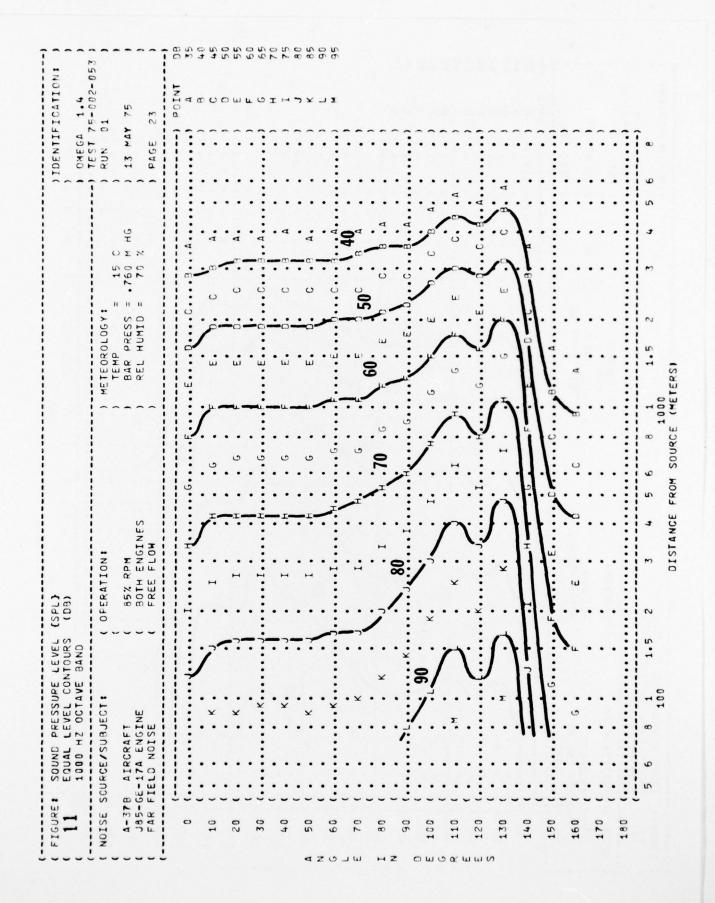


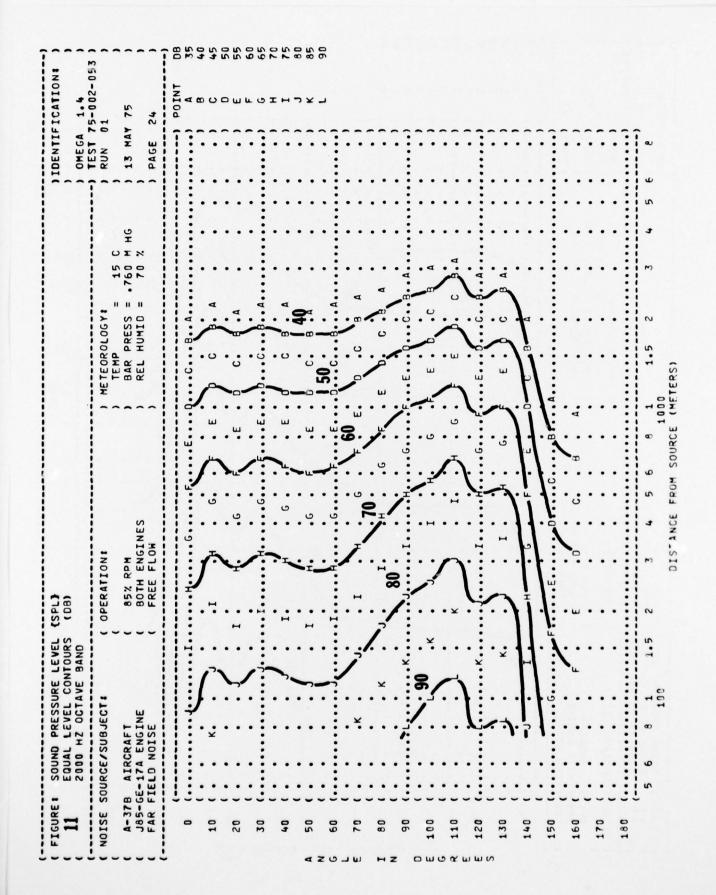


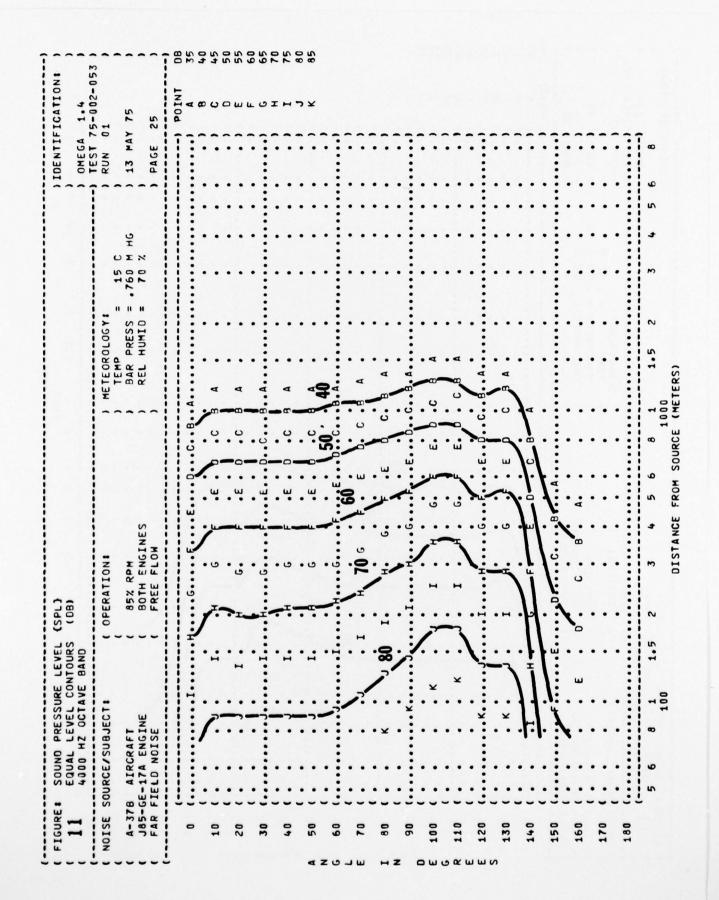




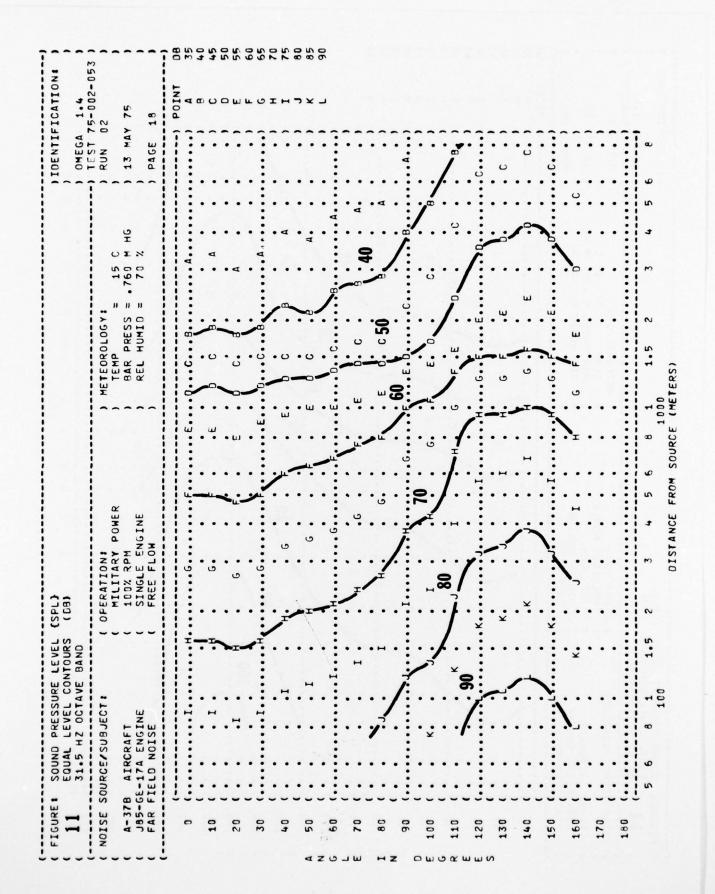


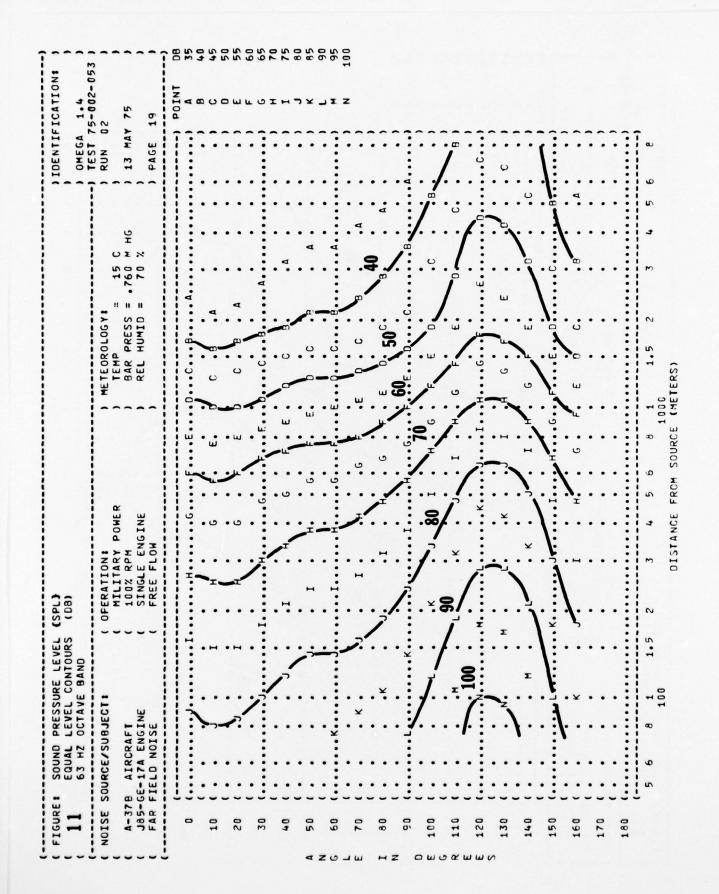


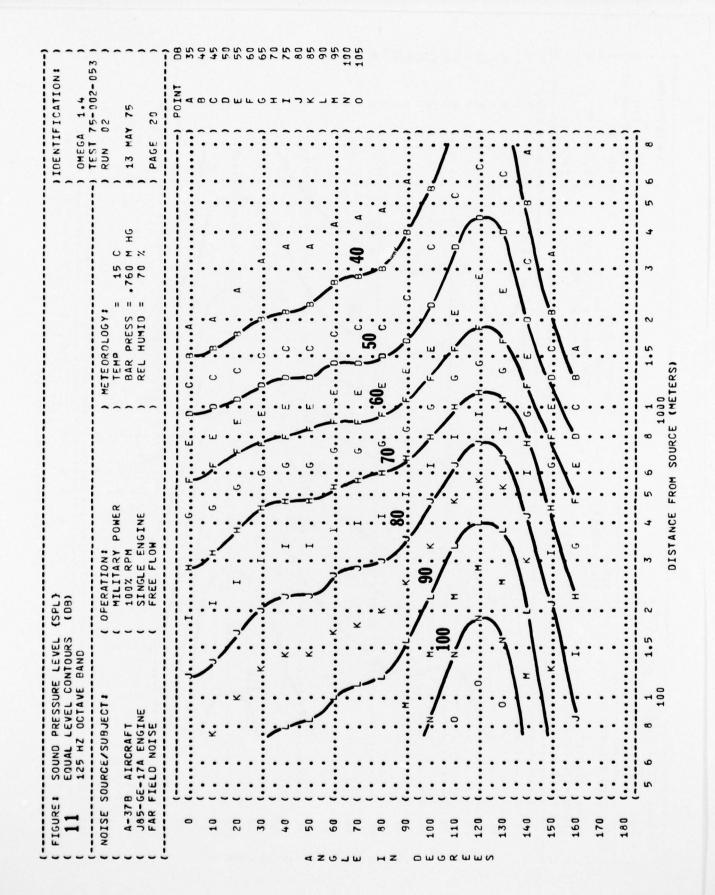


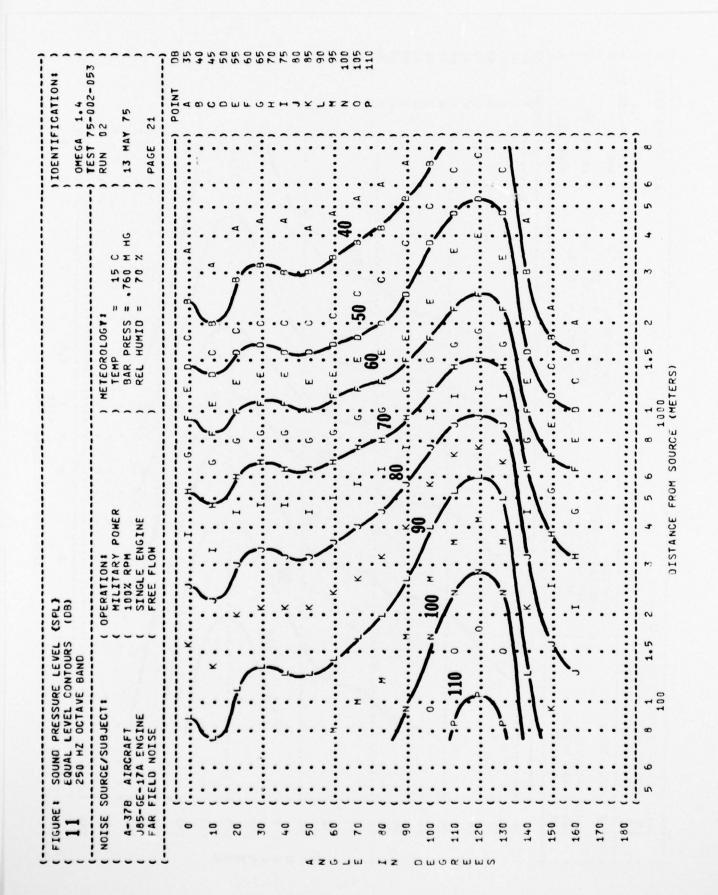


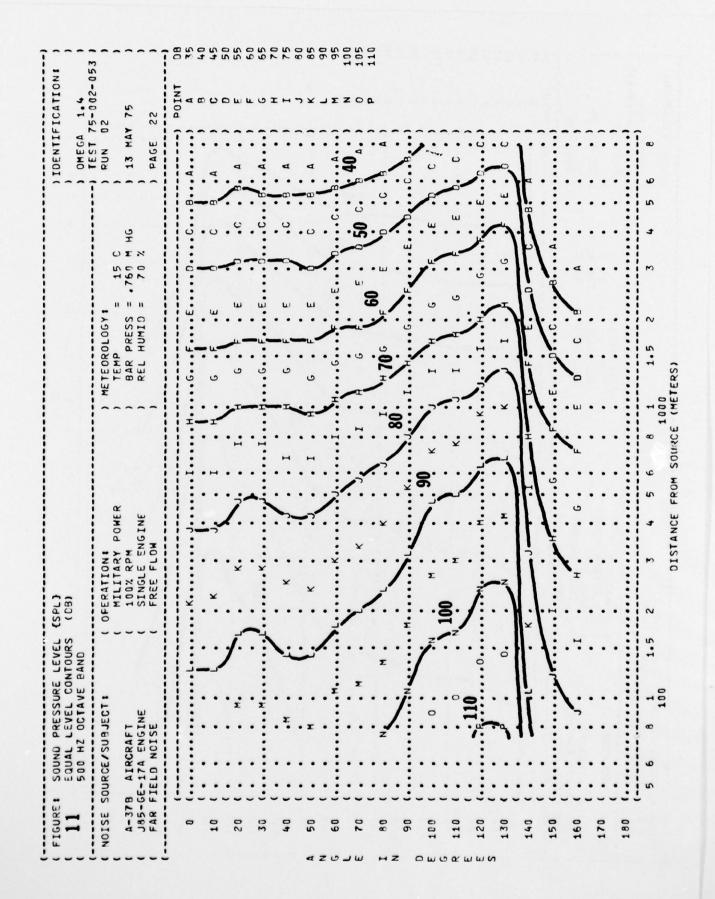
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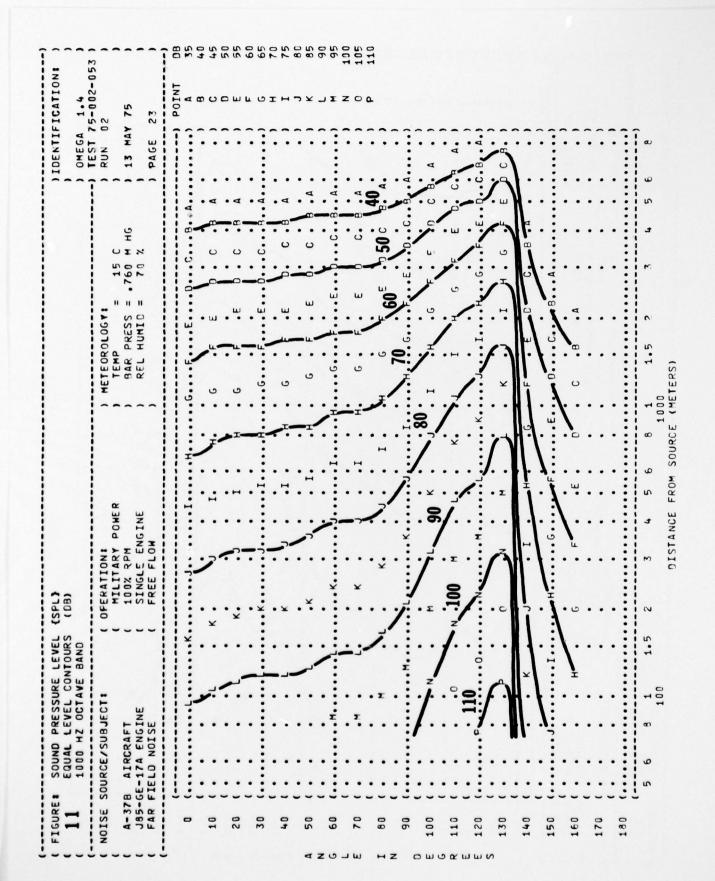


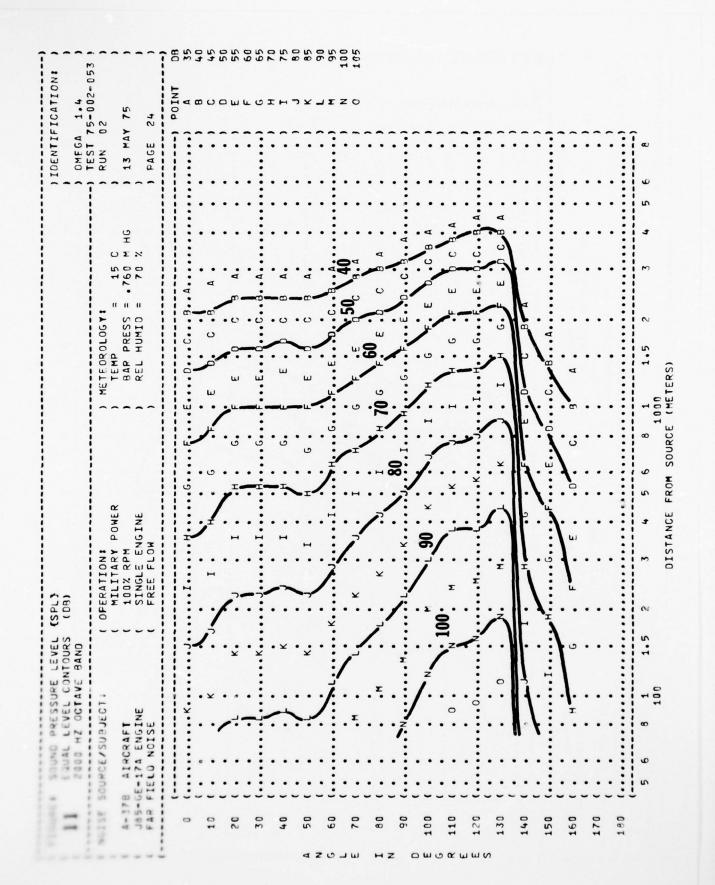


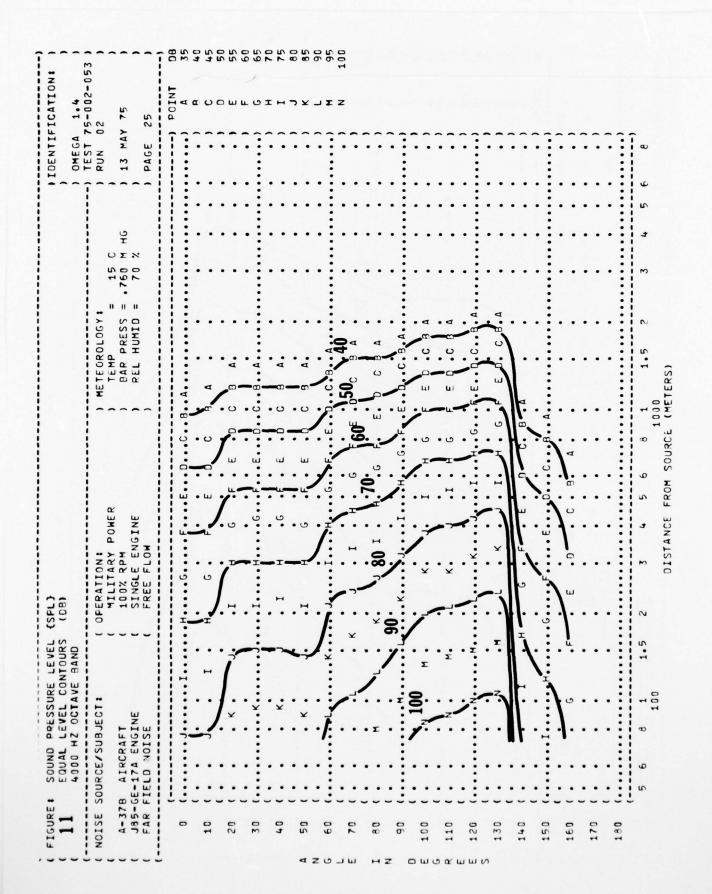












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